6. Design Procedure

6.1. Design for strength

Structural design codes give standardised design procedures for verifying the bending and lateral buckling strength, web shear strength and stability and connection strength. The design procedures stipulated in, AS 1418.18 should be regarded as complimentary with AS 4100 rules. With a view to minimise the design effort, AS 1418.18 categorises the runways into Light Duty for structural classes S1 to S7 and Heavy Duty for structural classes S8 and S9. The intention of the code committee was that full local detail checks for the top flange region are only necessary for Heavy Duty category runways. The author believes all crane runways should be checked.

There will be situations where a runway classified as S4 to S7 should be treated as the Heavy Duty category (heavy maintenance, steel mills, potroom service and general warehouse cranes). This is because the local stresses interact with the global stresses and this may lead to overstress.

AS 1418.18 standard covers some of the specialised design aspects of runway girders, such as: –

- Direct bearing stress distribution under the rail and in loaded flange to web junction.
- Local torsion in the upper flange region induced by lateral forces and wheel loads acting eccentrically with respect to the mid-plane of the web.
- Material type and thickness to avoid brittle fracture at low operating temperature. Already at +5 degrees C, plates in tension thicker than 20 mm should be of a steel exhibiting enhanced notch ductility, for example grades 300 -L0 and -L15, known as fully killed steels.
- Strict manufacturing and erection tolerances specified in the steel and runway codes must be achieved in practice because the code provisions are only valid for members fabricated and erected within those tolerances. Of special importance are the dimensional and alignment tolerances.
- Fatigue resistance should be considered in parallel with the detail design because in the majority of cases, the fatigue considerations govern the design of crane runways of structural classes S4 and over. It is best to specify details having the highest practicable fatigue detail category.
- Serviceability requirements such as deflections and twists should be checked against the specified limits as the design progresses.

Bending capacity is adequately covered in AS 4100. Interaction between bending and shear is covered in AS 4100, Section 5.12, which basically reduces the bending moment capacity when accompanied with relatively large shear forces. This starts to operate where M* is larger than $0.75M_s$.

6.2. Torsion

Torsion arises from lateral loads applied at the top of rail level and the eccentric application of vertical loads, that is, from the couples $N_w e_y$ and H h_t , see Figure 12(c). The eccentricity e_y has been specified in AS 1418.18 as consisting of two parts:

$$e_y = \frac{B_{te}}{k} + \frac{L}{1000}$$

where B_{te} is the effective railhead width,

k is an eccentricity factor;

k = 8 for rails with convex heads and

k = 4 for crane rails with flat head and rectangular bars.

This is a variation on the traditional $0.25B_{te}$ allowance.

The second term, L/1000 is for the lateral camber induced eccentricity based on tolerance for camber. AS 1418.18 allows the designer to neglect the second term when designing Light Duty runways. The author's opinion is that the camber induced eccentricity should be applied to all runways.

The torsion moment is a summation of ($N_{wi}^* e_y + H_{yt}^* h_T$) terms. Using the 'twin beam' analogy the lateral forces in flanges counteract the torsion moment alone and the contribution from pure torsion is neglected, see references 6, 55, 84 and 87.

Lateral forces acting on the top flange at each wheel (see Fig. 12) are given by

$$H_{ti}^{\star} = \frac{M_{Ti}^{\star} + N_{yi}^{\star} c_1}{h}$$

and the bottom flange forces:

$$H_{bi}^{\star} = \frac{-\ M_{Ti}^{\star} + \ N_{yi}^{\star} \, c_2}{h}$$

where M_{Ti}^{\star} = $N_{wi}^{\star}\,e_{y}$ + $N_{yi}^{\star}\,h_{t}$

These forces, applied laterally to the flanges, result in lateral bending moments, M_{yt}^* and M_{yb}^* as shown in Figure 12. It only remains to check the critical sections for combined actions.

$$\frac{M_{x}^{\star}}{\varphi M_{bx}} + \frac{M_{yt}^{\star}}{\varphi M_{sty}} \leq 1.0$$

where M_x^* is the major axis moment in vertical plane; M_{yt}^* is the lateral bending moment in the top (compression) flange; M_{bx} is the design moment capacity in vertical plane and M_{sty} is the lateral moment capacity of the top flange alone. M_{bx} is determined in accordance with member capacity of monosymmetric beams in AS 4100.

6.3. Torsion Capacity by rigorous method

Torsion induced by the lateral loads and rail contact eccentricity is not covered in AS 4100 at this stage. Rigorous methods of torsional analysis are described in references 49, 54, 55, 57, 80, 82, 84 but these tend to be cumbersome for practical application.

6.4. Lateral stability of the runway girder

Runway girders are normally restrained in the lateral direction at the supports only and usually have no intermediate lateral restraints. Intermediate lateral restraints can be beneficial but not easy to detail.

AS 4100 specifies an effective length factor of 1.40 where the loads are applied to the top of the girder flange. The fact that the loads are actually applied some distance above the top flange need not be considered because of the beneficial influence of linkage between the opposite side girders via the crane bridge. The verification of the lateral stability is to be carried out in accordance with AS 4100, as modified below.

The usual construction of light duty, low capacity runway girders is in form of a compound girder consisting of a UB section and Inverted top channel. This type of girder is termed monosymmetric girder. The warping constant used for symmetric and monosymmetric girders is given in AS 4100, Appendix H –

$$I_w = I_{cy} d_f^2 \left(1 - \frac{I_{cy}}{I_y} \right)$$

where

- I_{cy} = minor axis second moment of area of the top flange alone.
- I_y = minor axis second moment of area for the whole section.
- d_f = distance between centroids of flanges.

Recent research reported in Ref 88, Woolcock et al shows that AS 4100 method for monosymmetric girders can be unconservative. Neither method makes allowances for the beneficial effect of coupling via the crane bridge girders, such that the side carrying larger load receives some lateral support by the side under lower load.

As indicated in section 6.3 there is an interaction between the torsion and lateral buckling, and this is currently taken into consideration by using the effective length factor $k_e = 1.4$.

6.5. Box Sections

The method of stress analysis for box girder is described in Bennets and Grundy, ref 21. The dilemma is how to position the rail. Eccentric, over the web rail position is preferred in spite of imposition of large eccentricity. That should not be a problem because the box section has a relatively high resistance to torsion. The other option is to position the rail onto the centreline of the girder and cross stiffen the top plate at close centres. The latter method is costly and introduces a large number of fatigue prone welded connections.

The most important thing with box girders is the prevention of sectional distortion by means of diaphragm plates, stiff cross frames or cross bracing. The spacing of diaphragms can be as little as the depth of the girder or as much as one third span. Once diaphragms are provided the box girder will behave very much as an I-girder, but will have very much larger torsional stiffness.



Crane Runway Girders Limit States Design

Second Edition 2003





AUSTRALIAN STEEL INSTITUTE (ABN)/ACN (94) 000 973 839

Contents

1. Inti	roduction	1
2. Runway & Crane System		2
2.1. 2.2. 2.3. 2.4.	Crane types Crane runway girders Monorail beams Building columns and frames	3 5 7 8
3. Classification of Cranes and Runways		9
 3.1. 3.2. 3.3. 3.4. 3.5. 3.6. 	Reason for crane classification Utilisation Class – Global design Local Utilization class Multiple Cranes Structural Class v/s Group Class Duty Classification	9 9 10 10 11 11
4. Cra	ane Loads	12
 4.1. 4.2. 4.3. 4.4. 4.5. 4.6. 4.7. 4.8. 	Load determination	12 12 14 14 15 16 16
5. Structural Analysis		19
5.1. 5.2. 5.3. 5.4.	Global load effects Analysis for global loads Curved monorail beams Girder cross section	19 19 22 22
6. De	sign Procedure	24
6.1. 6.2. 6.3. 6.4. 6.5. 6.6. 6.7. 6.8. 6.9. 6.10	Design for strength Torsion Torsion Capacity by rigorous method Lateral stability of the runway girder Box Sections Design for fatigue resistance Local load effects in the top flange region Web stiffeners Lateral restraints at columns End stops	24 25 26 26 27 27 33 34 35
6.11	Monorail beams	36

7. Design for Fatigue Resistance	40
7.1. General	40 40
7.3. Number of stress cycles	41
7.4. Fatigue Verification by AS 4100	42
8. Deflection Limits	43
9. Detail Design	45
9.1. Detailing practices	45
9.2. Bolted connections	45
9.3. Welded joints	45
9.4. Splices in simply supported runways	46
9.5. Avoidance of lamellar tearing	46
9.6. Web stiffeners	47
9.7. End bearing stiffeners and bearing details	48
9.8. Crane columns and corbels	50
9.9. Longitudinal Bracing	51
10. Rails and Accessories	
10.1. Rail splices and expansion joints	52 53
10.3 Resilient bedding strips	54
10.4 Painting	54
	04
11. Materials, Fabrication, Workmanship and Tolerances	55
11.1. Materials	55
11.2. Workmanship	55
11.3. Welding top hat sections	56
11.4. Tolerances	56
12. Inspection and Maintenance	57
13. Numerical Example	58
14. Glossary	67
15. References	69

List of Figures

Fia 1.	Types of overhead running cranes	3
Fig 2.	Types of crane drives	4
Fig 3.	Runway Static System	6
Fig 4.	Monorail beam and cranes	7
Fia 5.	Relation between building frame and the runway	8
Fia 6.	Inertial forces	15
Fig 7.	Buffers and Buffer impact	16
Fia 8.	Oblique travel forces	17
Fig 9.	Crane wheel loads	18
Fig 10.	Frame / runway relation	19
Fig 11.	Bending moment envelope and influence lines	20
Fig 12.	Global analysis for vertical and torsional loads	21
Fig 13.	Curved monorail beam	22
Fig 14.	Types of cross section	23
Fig 15.	'Top Hat' (a) and lipped sections (b)	23
Fig 16.	Localized effects in the top flange area	27
Fig 17.	Web crushing (AS4100 method)	29
Fig 18.	Buckling of the web panel due to patch load	
-	acting in the plane of the web (AS 4100 method)	30
Fig 19.	Transverse bending of web due to torque	31
Fig 20.	Transverse bending of top flange	32
Fig 21.	Elastomeric strips reduce transverse bending of flange	33
Fig 22.	Web stiffeners	34
Fig 23.	Lateral movement and rotation at girder bearing	35
Fig 24.	Forces on end stops	36
Fig 25.	Monorail bottom flange stresses	37
Fig 26.	Comparison of Becker(15) vs BHP plots for Cz under the wheel load	38
Fig 27.	Stress range vs number of stress cycles for normal stress and shear stresses (excerpt from AS4100)	42
Fig 28.	Deflection limits	44
Fig 29.	Rail meandering in practice	44
Fig 30.	Bolted intermediate web stiffeners	45
Fig 31.	Welded girder splices	46
Fig 32.	Web stiffener details	47
Fig 33.	End bearing stiffeners	48
Fig 34.	Bearings	49
Fig 35.	Unsatisfactory bearing details	49
Fig 36.	Types of supporting columns	50
Fig 37.	Longitudinal Expansion due to temperature and bracing of crane columns.	51
Fig 38.	More bearings and crane rail splice details	53
Fig 39.	Common rail fixings	54
Fig 40.	Soft bedding of rails	54
Fig 41.	Welding access to 'top hat' welds	56
Fig 42.	Applied Loads	58
Fig 43.	Section and Girder Dimensions	58
Fig 44.	Forces from trolley acceleration	59